



Empirical Modeling of Solar Radiation Pressure Forces Affecting GPS Satellites

Ant Sibthorpe, Jan Weiss, Nate Harvey,
Da Kuang and Yoaz Bar-Sever

Jet Propulsion Laboratory,
California Institute of Technology

AGU Fall Meeting, San Francisco

Dec 17, 2010



Outline



- Motivation
 - Empirical Solar Radiation Pressure Modeling
 - Internal Metrics
 - Ambiguity Resolution, Orbit Overlaps
 - External Metrics
 - Satellite Laser Ranging
 - Orbit Prediction
 - Conclusion
- June 2009
to
June 2010
- June - Nov 2010



Terminology



Precise Orbit Determination (POD)

Solar Radiation Pressure (SRP)

- **SRP Model (today's focus!):**

A priori representation of SRP forces acting on a spacecraft

- **SRP Strategy:**

Approach to estimating SRP forces during POD

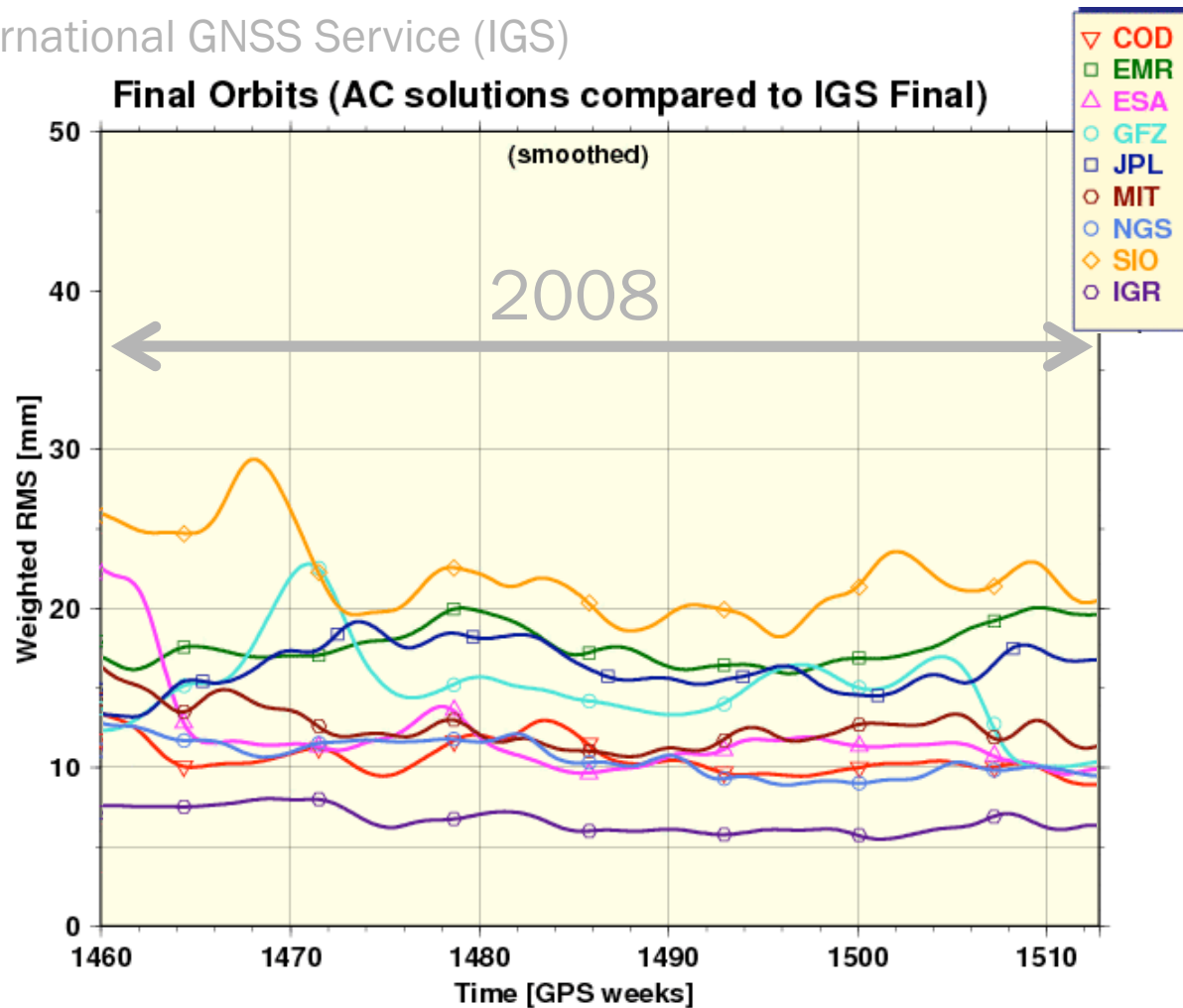
Does not explicitly require an a priori SRP model



Motivation



International GNSS Service (IGS)



<http://acc.igs.org/>

NOAA NGS, 24.01.2009 19:03 (GMT)



GSPM04



- GPS Solar Pressure Model - Bar-Sever & Kuang (2004, 2005)
- Based on 4.5 years of precise orbits (Jan 1998 – June 2002)
- Truncated Fourier series - coefficients combined from 10-day fits
- β -angle dependent SX_2 and CY_1 terms
- Block specific models (IIA/IIR)

$$F_x = \frac{k10^{-5}(AU/r)^2}{m \sum_{i=1,2,3,5,7} SX_i \sin(iEPS)}$$

$$F_y = \frac{CY_0 + 10^{-5}(AU/r)^2}{m \sum_{i=1,2} CY_i \cos(iEPS)}$$

$$F_z = \frac{k10^{-5}(AU/r)^2}{m \sum_{i=1,3,5} CZ_i \cos(iEPS)}$$

k dimensionless scale factor
m spacecraft mass (kg)
EPS Earth-Probe-Sun angle



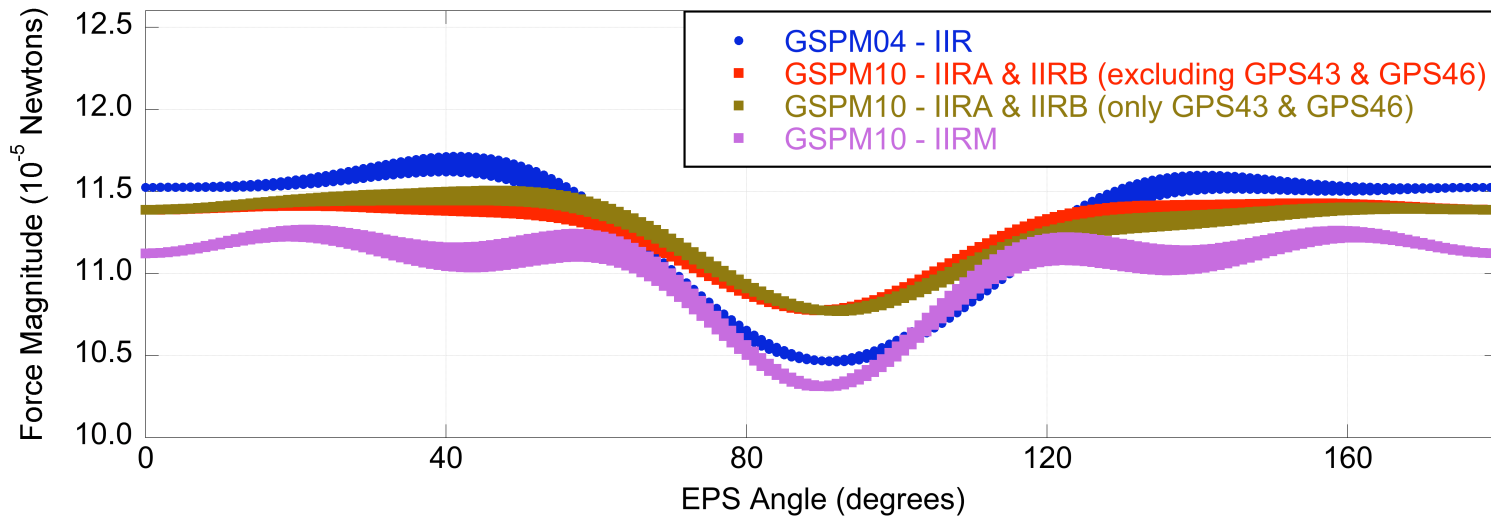
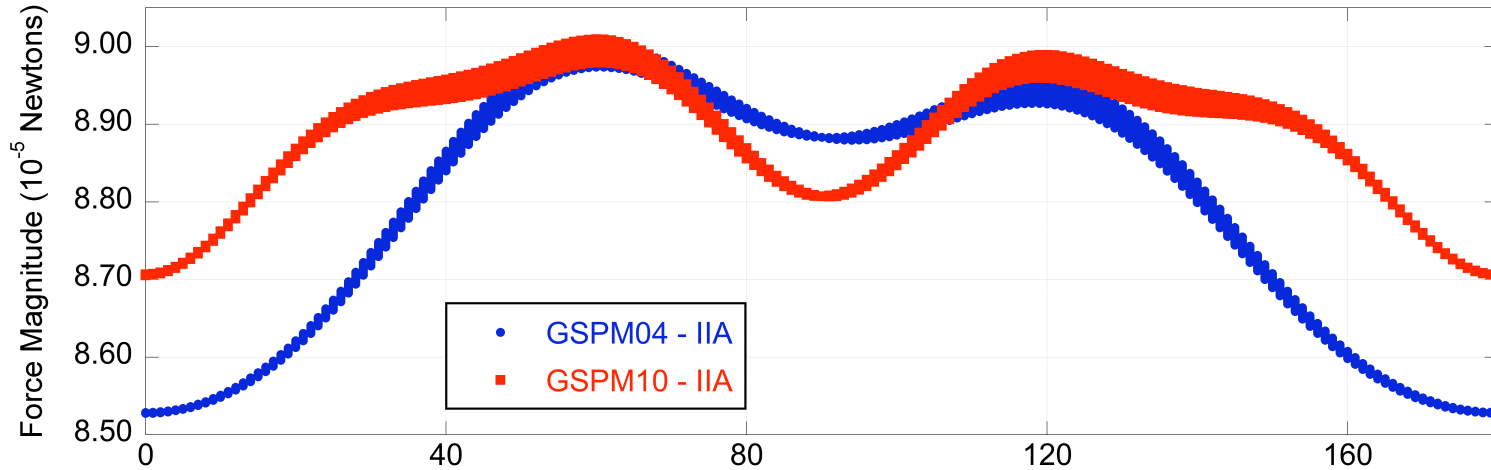
GSPM10



- GNSS Solar Pressure Model (GPS & GLONASS)
- More rigorous combination of coefficients from 10-day fits
- Improved modeling of β -angle dependent terms SX_2 and CY_1
- For GPS - 13.5 years of precise orbits (Jan 1997 – May 2010)
- Separate GPS models for:
 - I/A (non-eclipsing only)
 - IIR-A & IIR-B (excluding GPS43 & 46)
 - IIR-A & IIR-B (GPS43 & GPS46 only)
 - IIR-M



Model Comparison





POD Results



Daily orbit & clock solutions computed from:

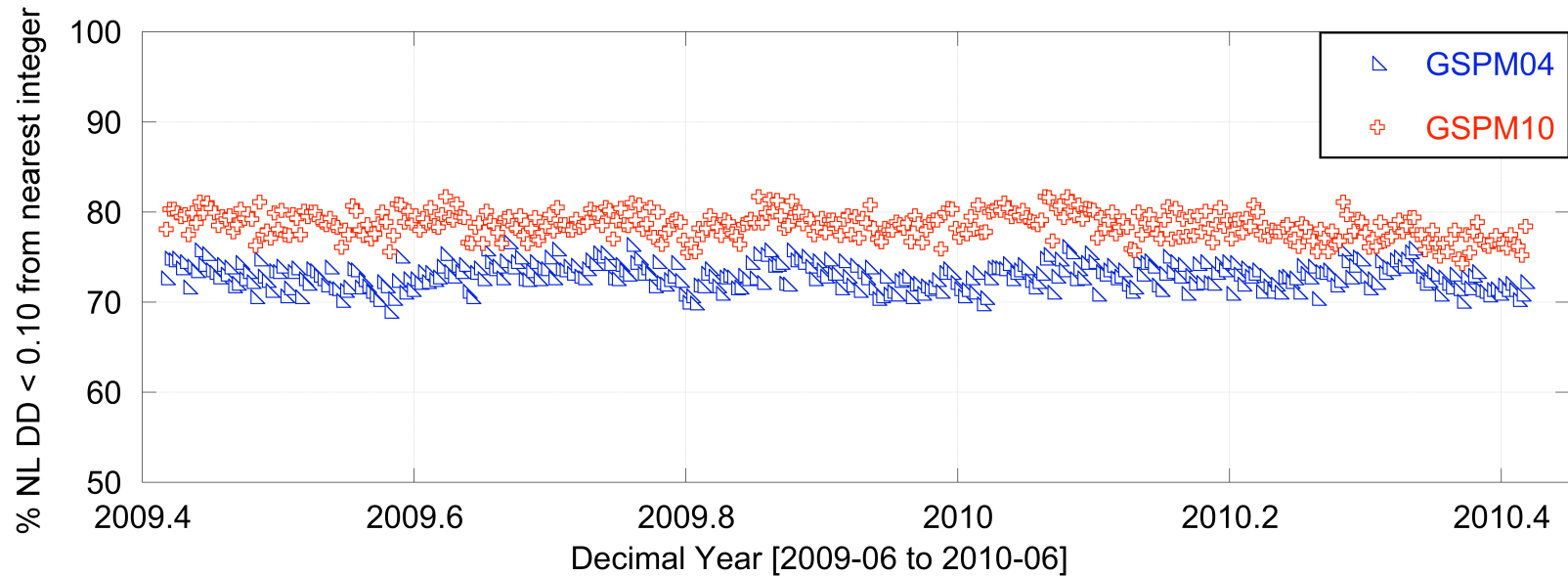
- GSPM04
- GSPM10

across the period June 2009 to June 2010.

Identical input data, software (GIPSY-OASIS) and strategies



Ambiguity Resolution

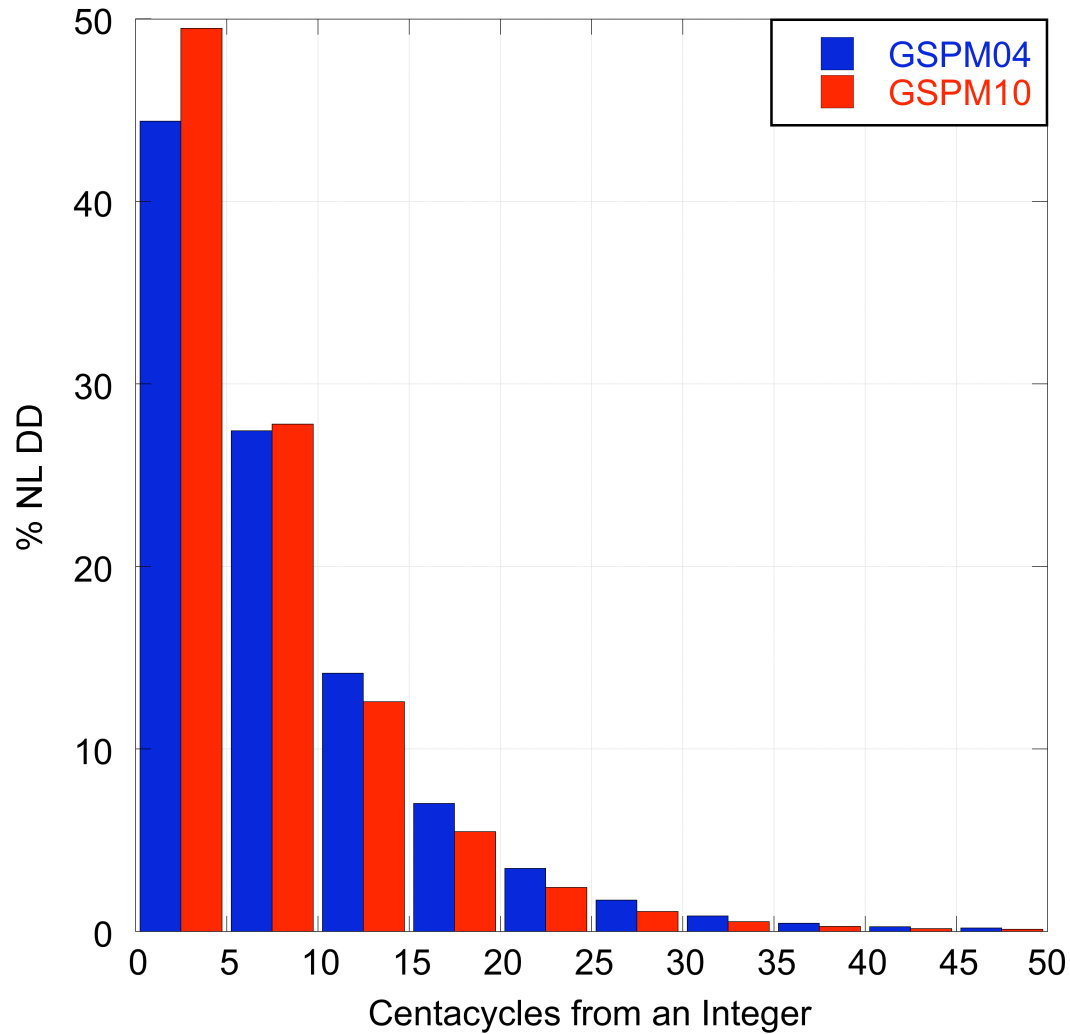


	Mean	1- σ
GSPM04	73.0	1.4
GSPM10	78.6	1.4

Values in %



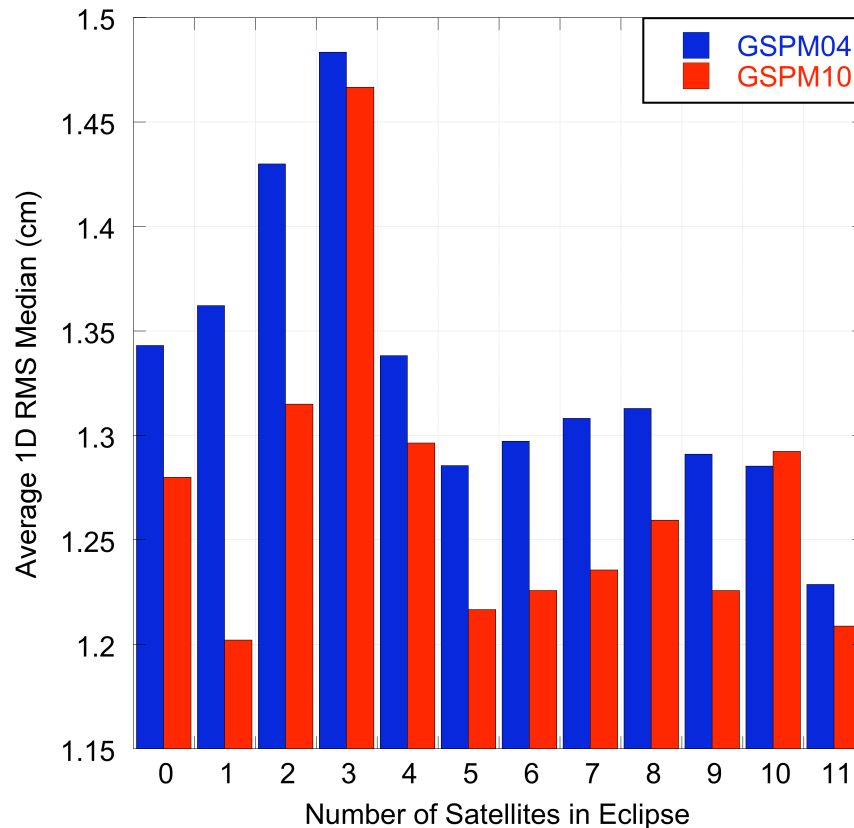
Ambiguity Resolution



	# to fix
GSPM04	55836681
GSPM10	55828557



Orbit Overlaps



Maximum (across available satellites) 1D RMS values of day-to-day overlap differences.

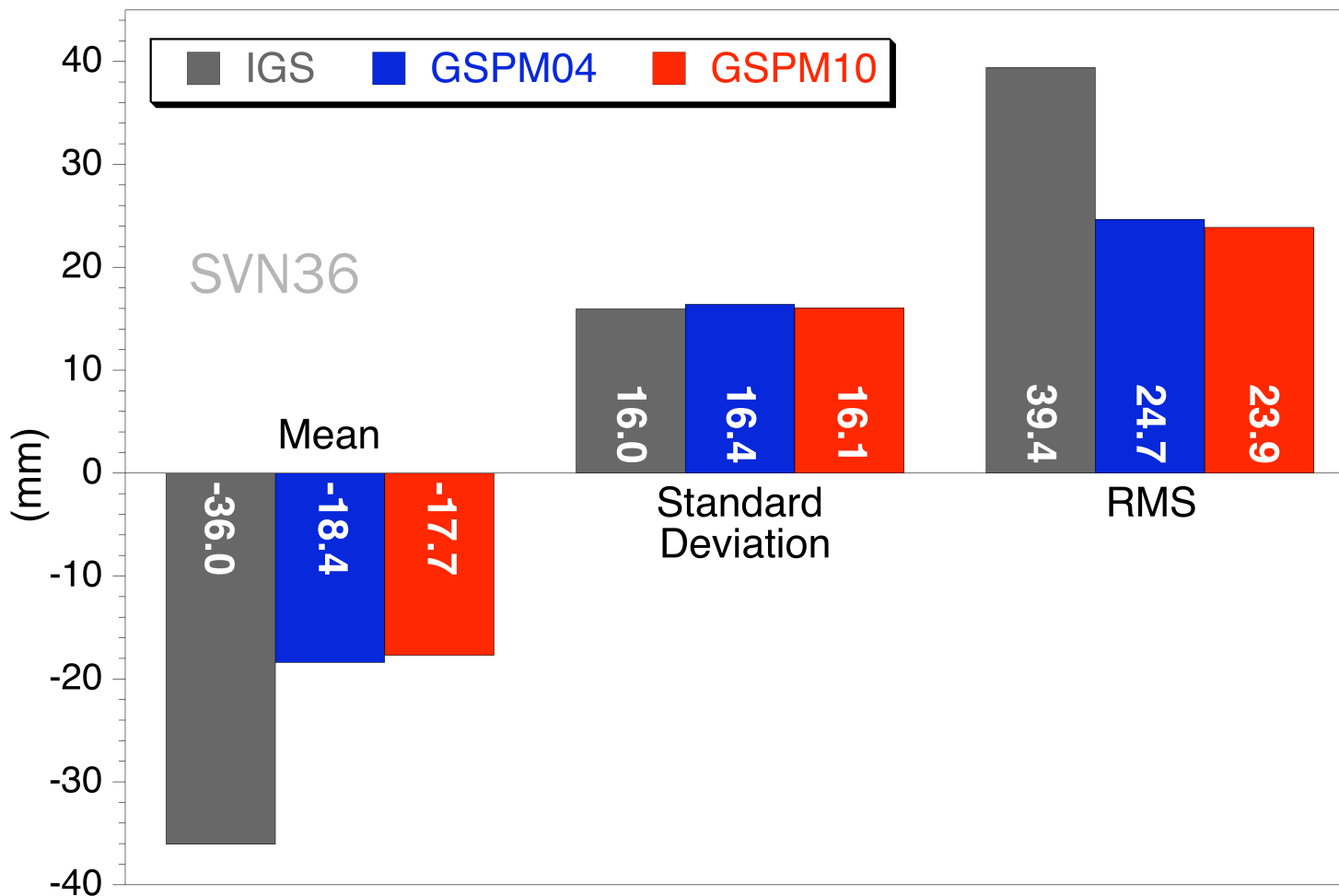
Median (across available satellites) 1D RMS values of day-to-day overlap differences.

Non-eclipsing satellites only	GSPM04	GSPM10
	Mean	Mean
Maximum	2.33	2.28
Median	1.31	1.25

Values in cm

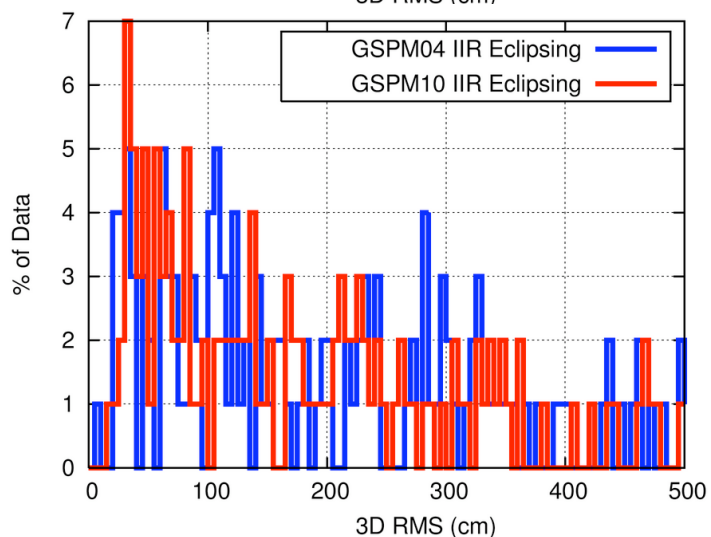
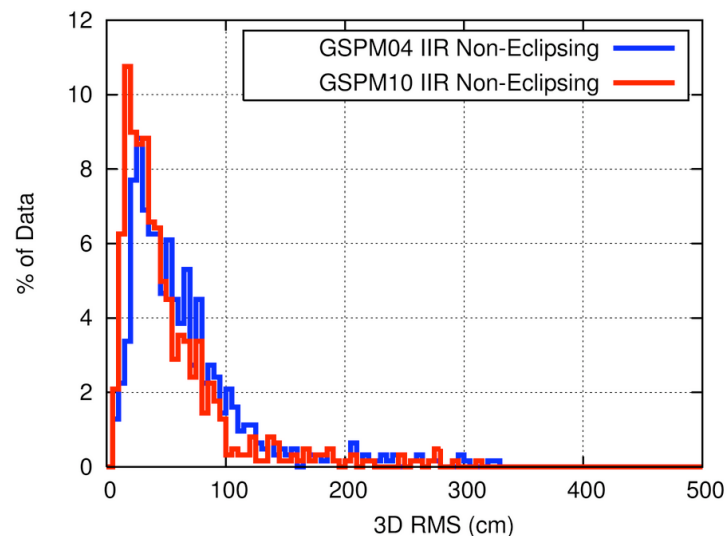
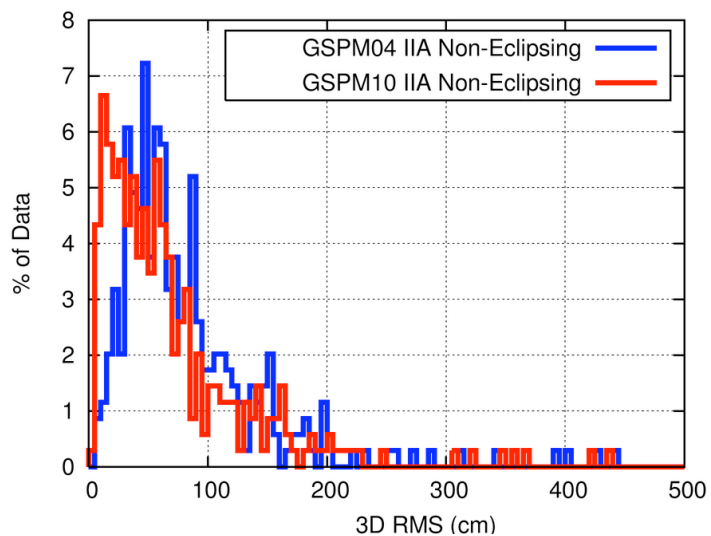


Satellite Laser Ranging





Orbit Prediction



4th day prediction-error 3D RMS

	GSPM04	GSPM10	% Improved
	Mean	Mean	
IIA Non-Eclipsing	73.6	57.8	21.5
IIR Non-Eclipsing	52.5	40.3	23.2
IIR Eclipsing	250.7	233.0	7.1

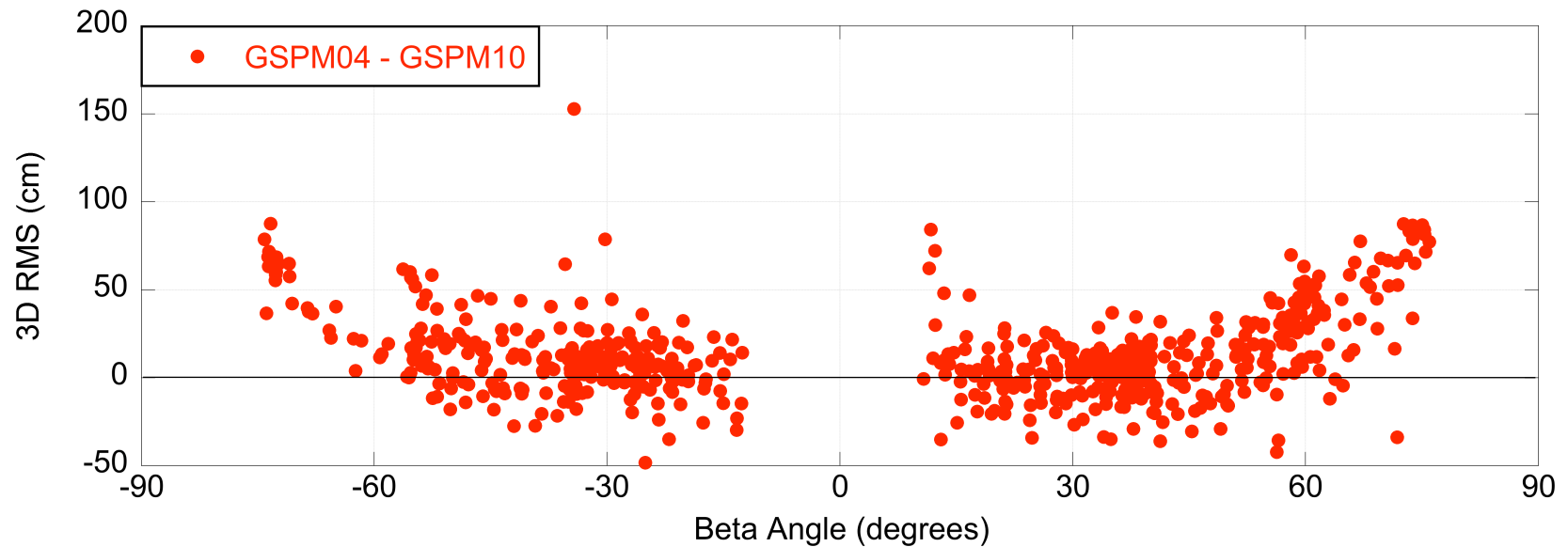
Values in cm



Orbit Prediction



Non-eclipsing GPS IIR 4th day prediction error difference by beta angle (GSPM04 - GSPM10). Positive values indicate improvement from GSPM10.





Conclusions



- GSPM10 - 14 years of data, models for IIA, IIR-A/B, IIR-M
- Assessing orbit accuracy is becoming increasingly demanding
- Orbit overlaps and satellite laser ranging are capable of differentiating the GSPM solutions
- GSPM10 model performance strongly supported in GIPSY by:
 - Ambiguity resolution statistics
 - Orbit prediction (better sampling and β -angle dependency)



Future Work



- IIA eclipsing model
- Iterate as more data becomes available
- IIF
- GSPM10 update for GLONASS due to new attitude model (Weiss et al., AGU 2010, poster)